

# DEVELOPMENT OF THE COUPLING BETWEEN DPLR AND NEQAIR

Entry System and Technology Division  
NASA Ames Research Center  
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Chiara Amato

Mentors: Joseph Schulz and Aaron Brandis

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- 1 Motivations
- 2 Fluid-Radiative Coupling
- 3 Results
- 4 Summary & Future Work



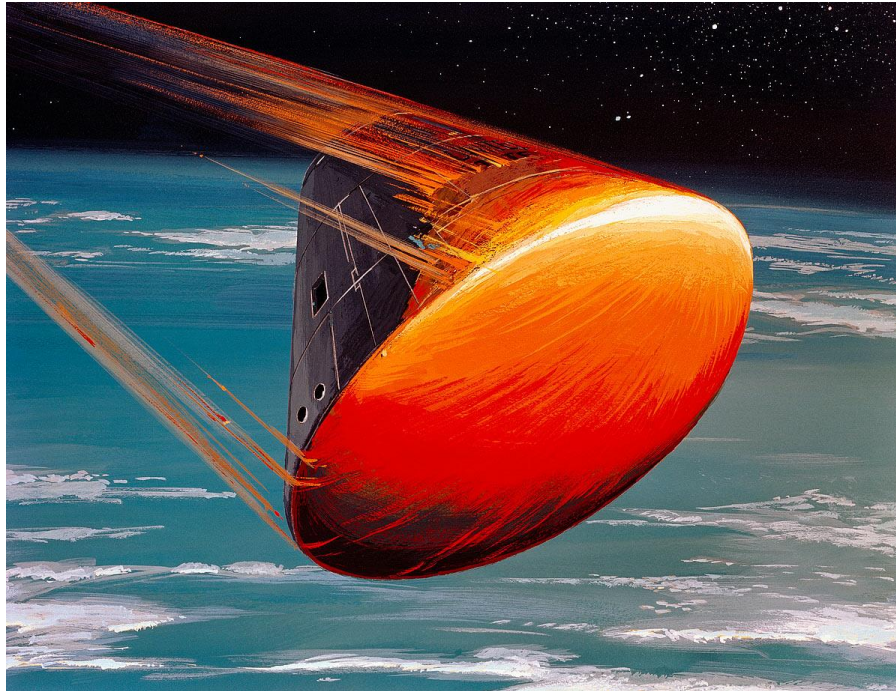
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# MOTIVATIONS

- The heatshield of a spacecraft entering an atmosphere at high velocities will experience high heat loads.
- The heat transfer to the spacecraft is one the most important parameters for Thermal Protection System (TPS) design and one of the most difficult to calculate numerically.
- TPS design requires an accurate prediction of the heating rates (material selection) and the total integrated heat load (material thickness).



# MOTIVATIONS

- Consider two components of the energy balance: **CONVECTIVE** and **RADIATIVE** heating;
  - radiant energy can be a significant fraction of the total energy
- Including radiative heating can:
  - Reduce bow shock standoff distance
  - Reduce the total surface heat load through radiative cooling

## ENERGY BALANCE

$$\frac{\partial e}{\partial t} + \frac{\partial}{\partial x_j} \left( (e + p)u_j \right) = -\Theta^R - \frac{\partial}{\partial x_j} (q + q_{vj} + q_{ej}) - \frac{\partial}{\partial x_j} (u_i \tau_{ij}) - \frac{\partial}{\partial x_j} (u_{si} (h_s \delta_{ij} + \tau_{sij}))$$

Convective Energy Term

Radiant Energy Source Term

Viscous flux terms

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# FLUID – RADIATION COUPLING

- Data-Parallel Line Relaxation (DPLR) Code:
  - 2-D/3-D nonequilibrium Navier–Stokes flow solver,
  - accurate analysis of the flow under hypersonic conditions;
- The NEQAIR code is a line-by-line radiation code:
  - radiation computation along each line of sight;
  - flow information provided by DPLR;
- The Coupler code is a Python interface:
  - sets up all the directories and files for a coupled calculation;
  - creates the required couple tools.



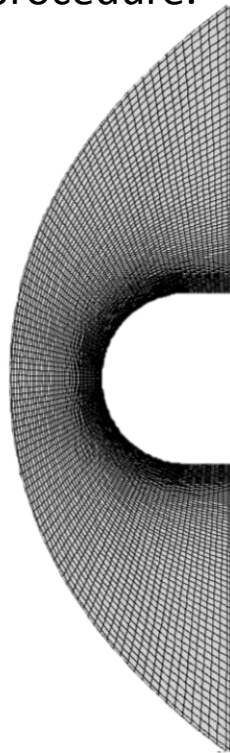
# Fluid – Radiation Coupling

Iteration 0

Uncoupled DPLR  
run

Free stream  
→

- Grid alignment necessary for accurate CFD analysis:
  - reduces number of grid cells from the free stream
  - adjusts the outer boundary to align with the shock wave
  - decreases wall spacing to resolve boundary layer
- A simple geometry was analyzed to get familiar with the grid alignment procedure.



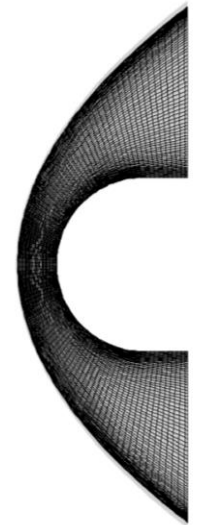
No alignment



1<sup>st</sup> alignment



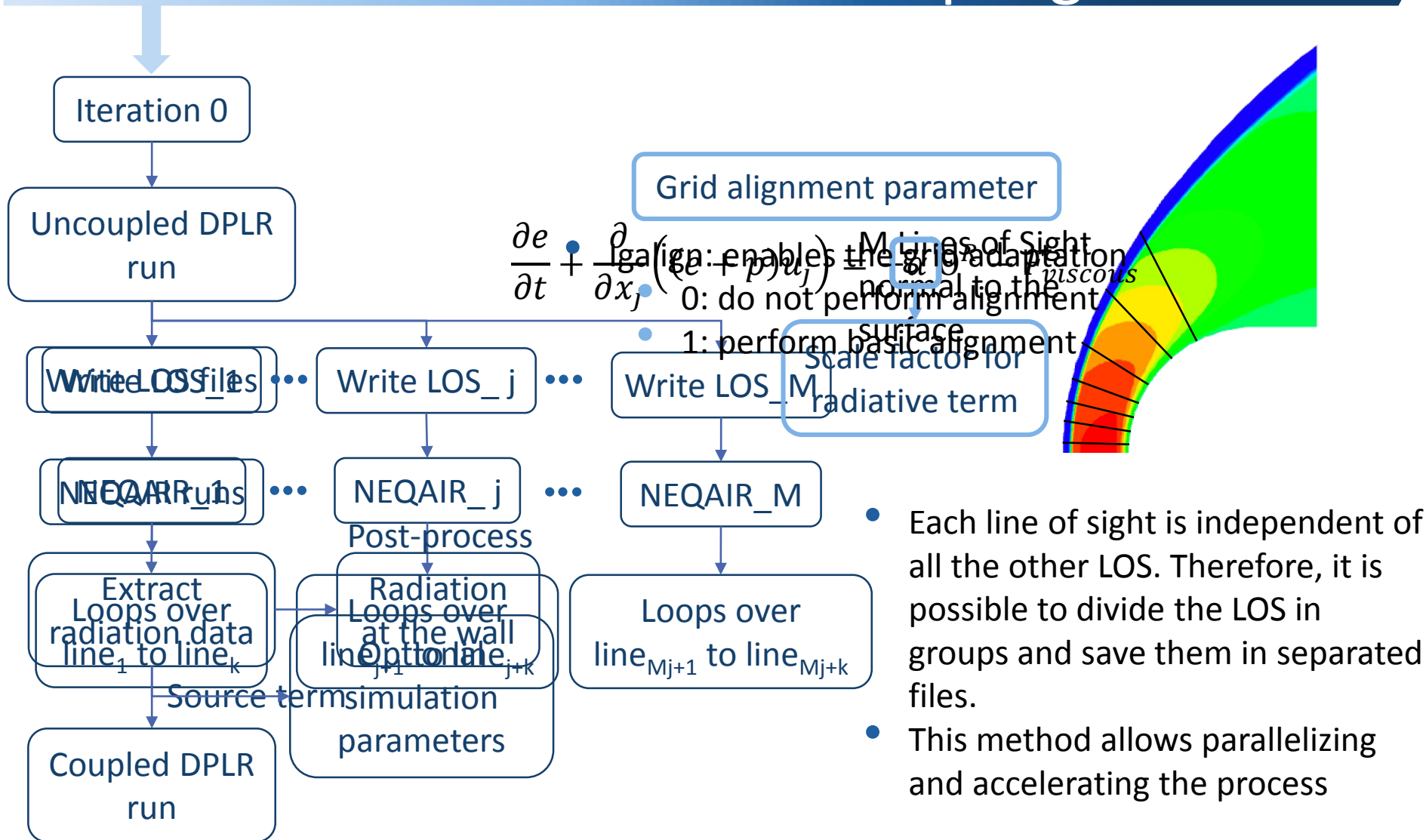
2<sup>nd</sup> alignment



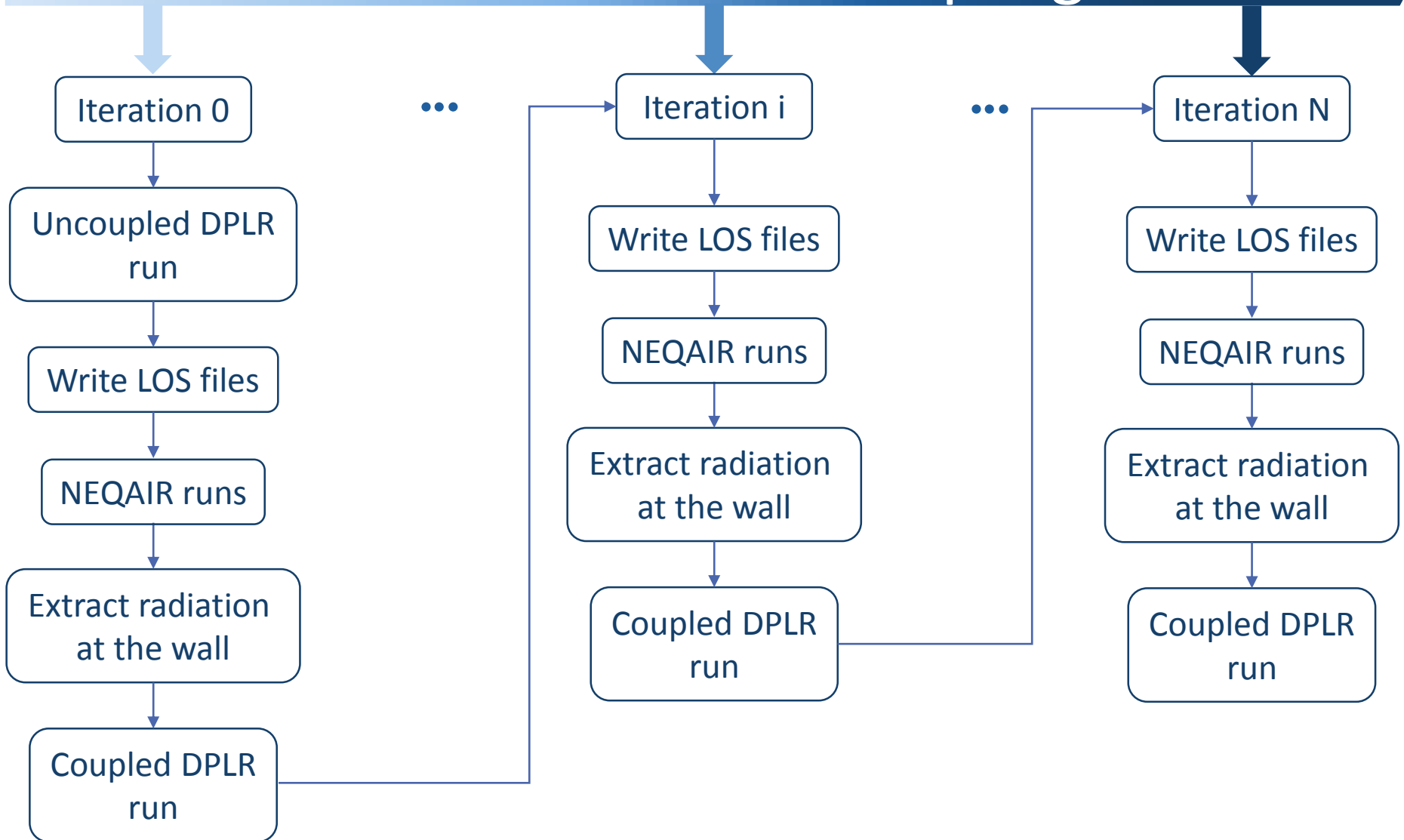
3<sup>rd</sup> alignment



# Fluid – Radiation Coupling



# Fluid – Radiation Coupling



# COUPLER INPUT FILE

```
[INPUT_FILES]
!-----!
dplr_dir = /User_dir/DPLR-NEQAIR/testcase/dplr
dplr_input = dplr.inp
dplr_post = post.inp
dplr_grid = grid.pgrx
dplr_flow = flow_sol.pslx
neqair_dir = /User_dir/DPLR-NEQAIR/testcase/neqair
neqair_input = neqair.inp
!-----!
[/INPUT_FILES]

dplr_runs_per_iteration = 1
coupling_run_dirs = iter_
dplr_run_dirs = dplr_
neqair_run_dirs = neqair_
dimensions = 2
normal_line = j
!-----!

[DPLR_OPTIONS]
!-----!
scale_divq = [0.2, 0.5, 0.75, 1.0]
!-----!
[/DPLR_OPTIONS]

nodes = 1
cores = 64
days = 0
hours = 2
minutes = 0
jobname = test2couple
neqair_mpi_procs = 64
neqair_log = neqair-log
neqair_runs = 4
neqair_batch_mode = independent
dplr_mpi_procs = 32
dplr_log = dplr-log
!-----!
[/BATCH]
```

Input directories for  
uncoupled flow solution and  
NEQAIR input file

Run directories numbered by  
prefix and iteration count.  
Separate directories created  
for DPLR and NEQAIR runs.

Scale factor for the  
radiative term

Number of NEQAIR and  
LOS iterations, and  
SLURM variables.

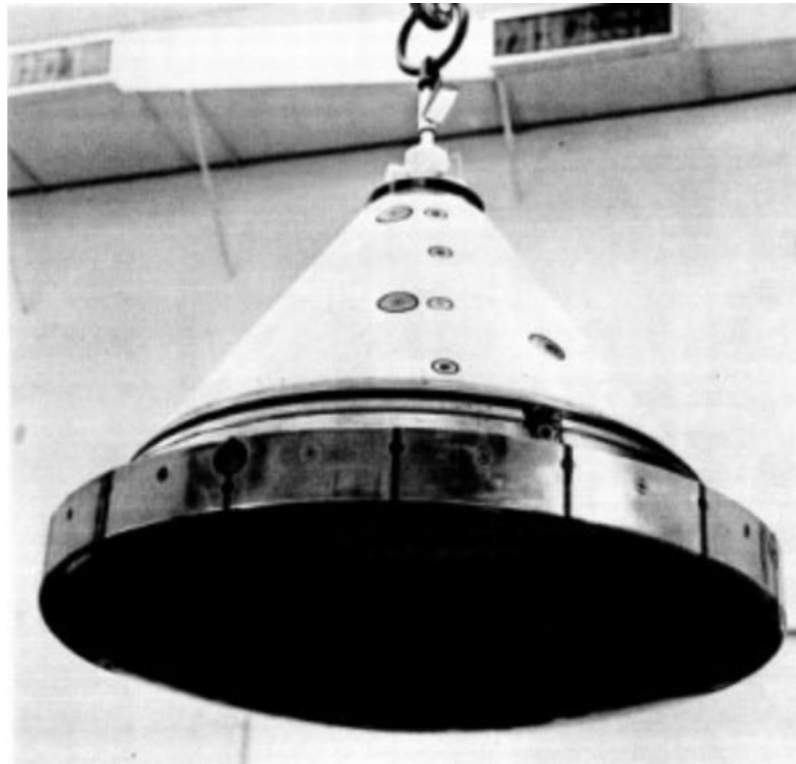
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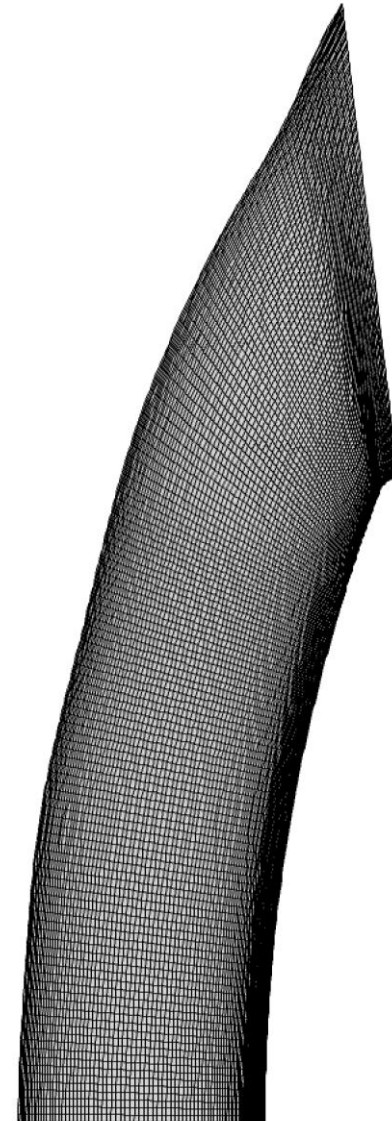
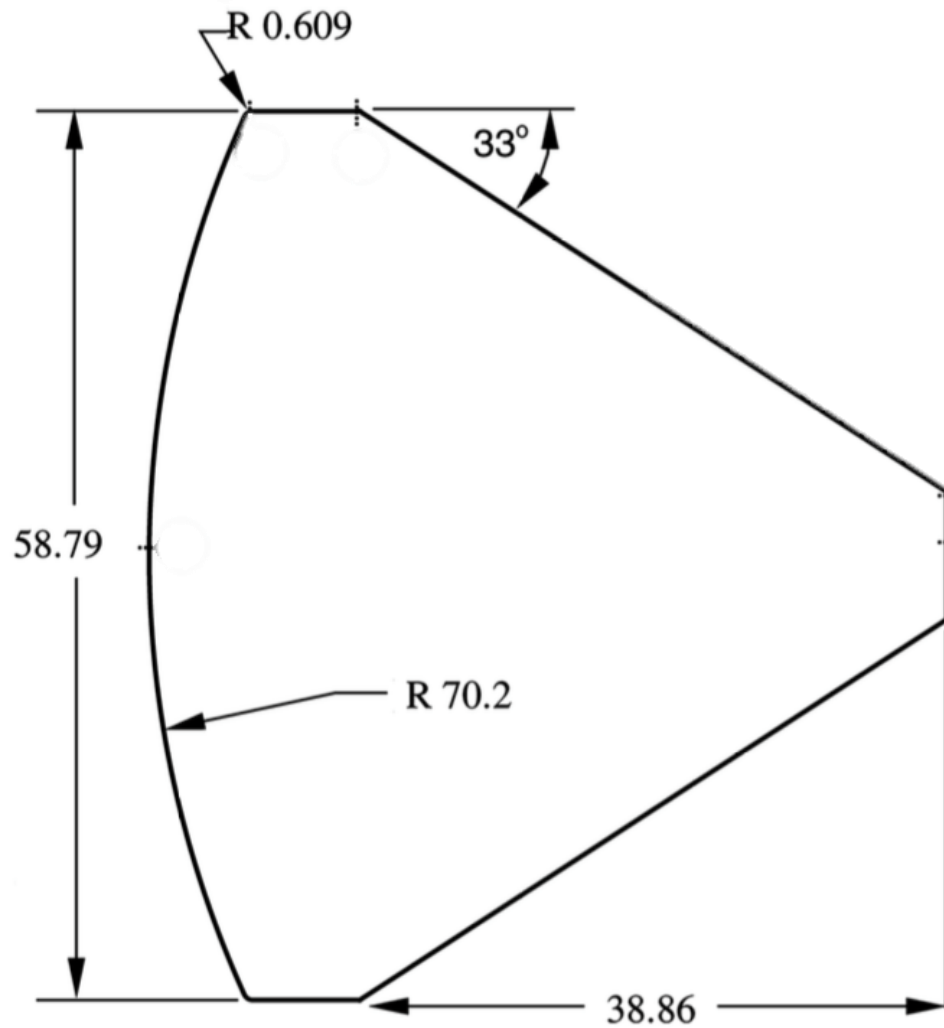


# FIRE II CAPSULE

- Project FIRE (Flight Investigation of the Reentry Environment) , which flew in 1965, was conducted expressly for the purpose of measuring radiative heating at Lunar return velocities.
- The FIRE-II experiment is considered to be the best documented data set available on radiative heating.



# FIRE II GEOMETRY



# FIRE II CONDITIONS

## DPLR

- Two-temperature model describes the thermochemical nonequilibrium:
  - translational-rotational temperature  $T$
  - vibrational-electronic temperature  $T_v$ ;
- An 11 species gas model is used to simulate the flow:  $N_2$ ,  $O_2$ ,  $NO$ ,  $NO^+$ ,  $N_2^+$ ,  $O_2^+$ ,  $N$ ,  $O$ ,  $N^+$ ,  $O^+$ ,  $e^-$ ;
- Park's 1993 21-reaction, 11-species kinetic model.

## NEQAIR

- The calculations were performed using a Boltzmann distribution assumption;
- The spectral range was analyzed over 85.5-3960 nm;

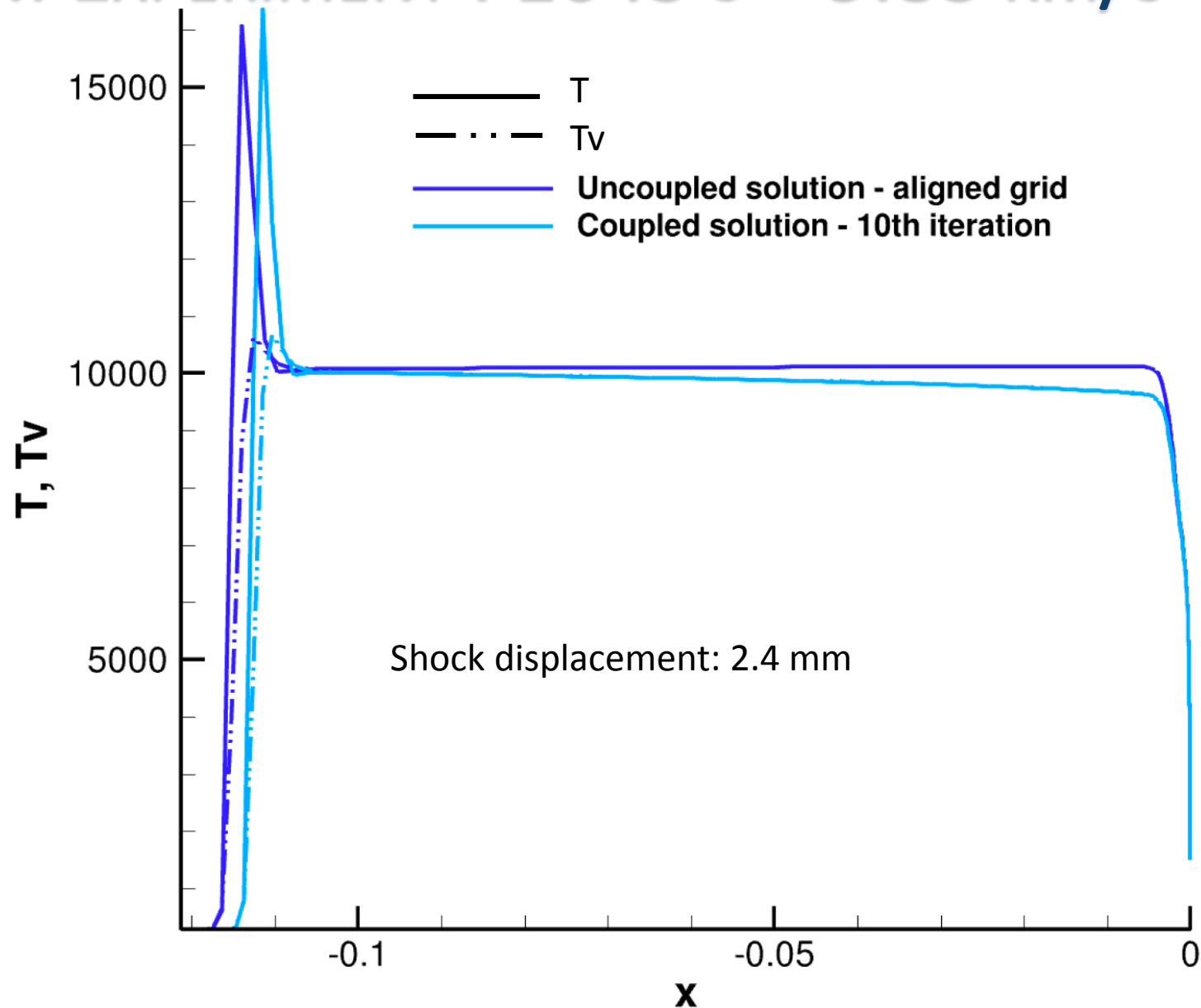


# FREE STREAM CONDITIONS

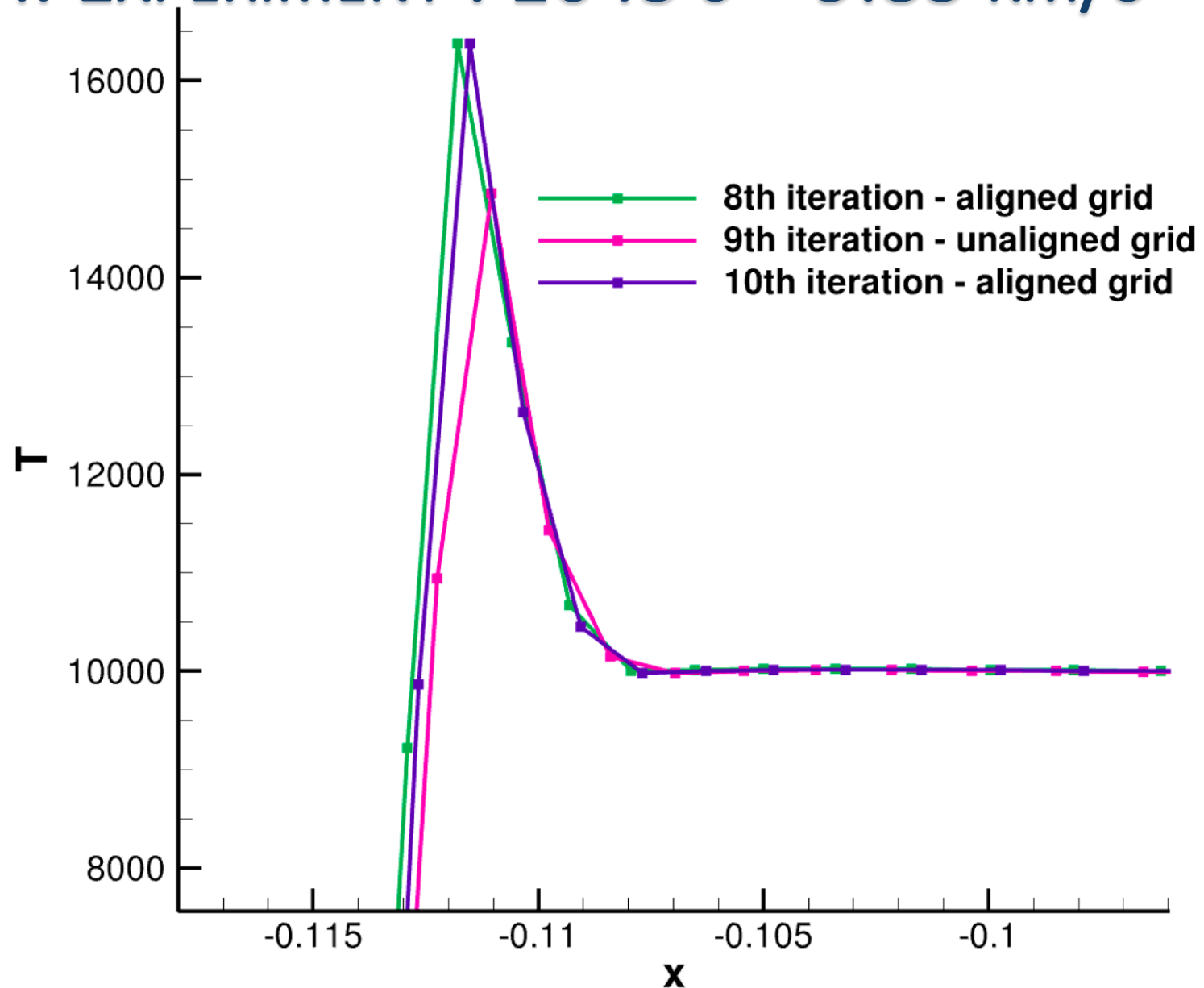
Experiment	Free Stream Entry Velocity [km/s]	Free Stream Density [kg/m <sup>3</sup> ]	Free Stream Translational Temperature [K]	Wall temperature [K]
FIRE II 1645 s	9.83	$1.32 \cdot 10^{-3}$	285	1520
FIRE II 1643 s	10.48	$7.8 \cdot 10^{-4}$	276	640
FIRE II 1636 s	11.3	$8.75 \cdot 10^{-5}$	210	810

Lewis Jr, John H., and William I. Scallion. "Flight parameters and vehicle performance for Project Fire flight II, launched May 22, 1965." (1966). NASA Langley Research Center

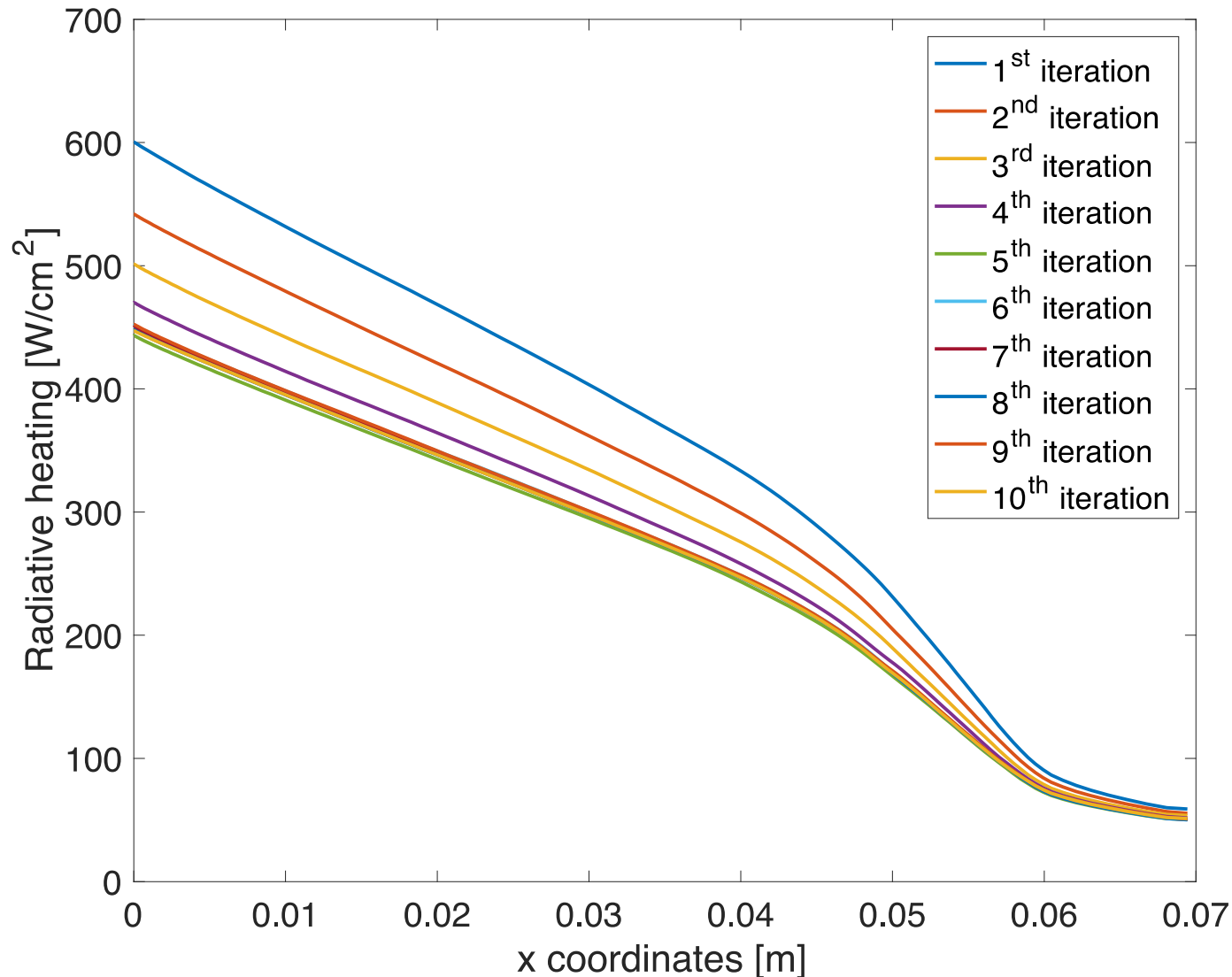
# FIRE II EXPERIMENT : 1645 s – 9.83 km/s



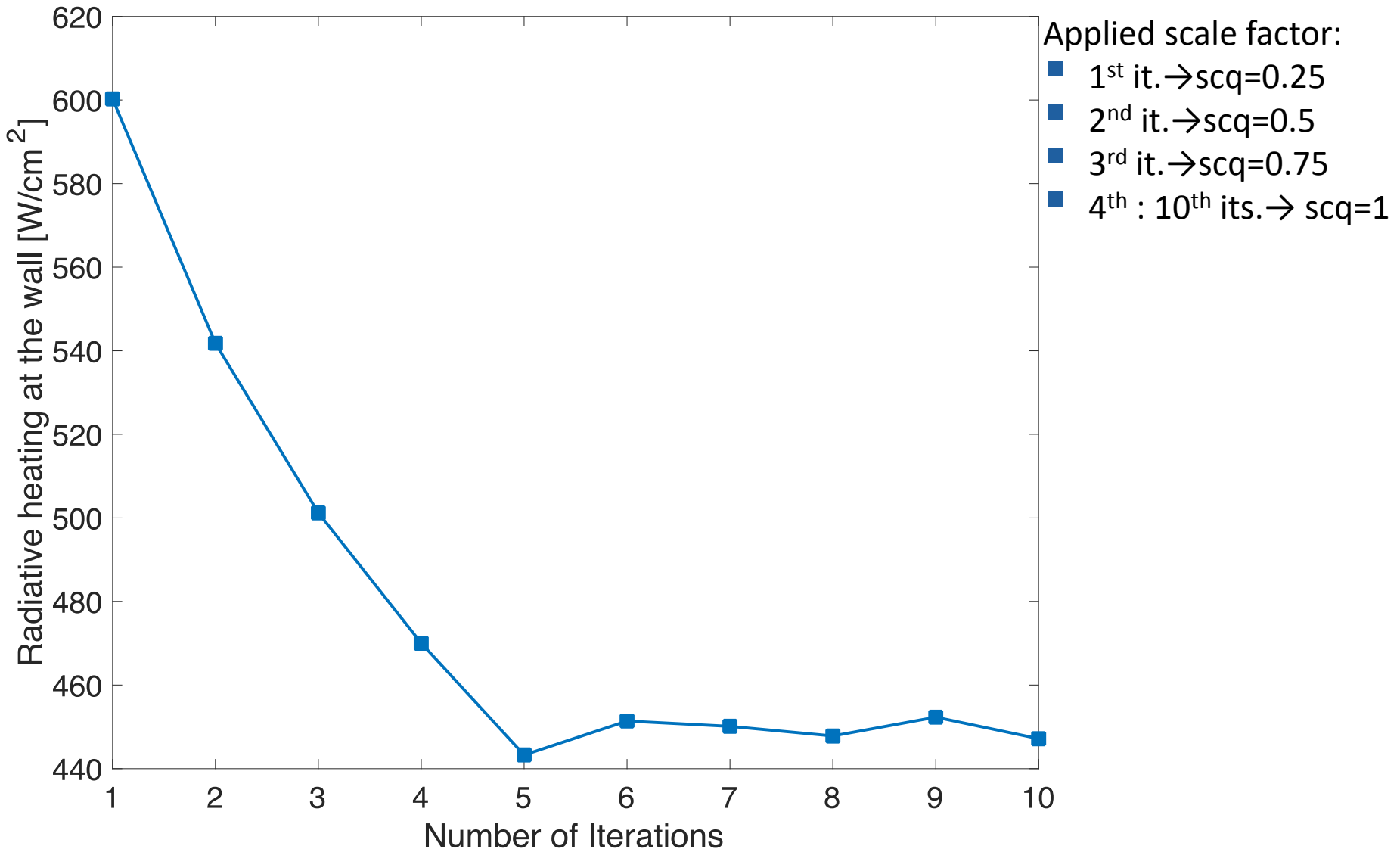
# FIRE II EXPERIMENT : 1645 s – 9.83 km/s



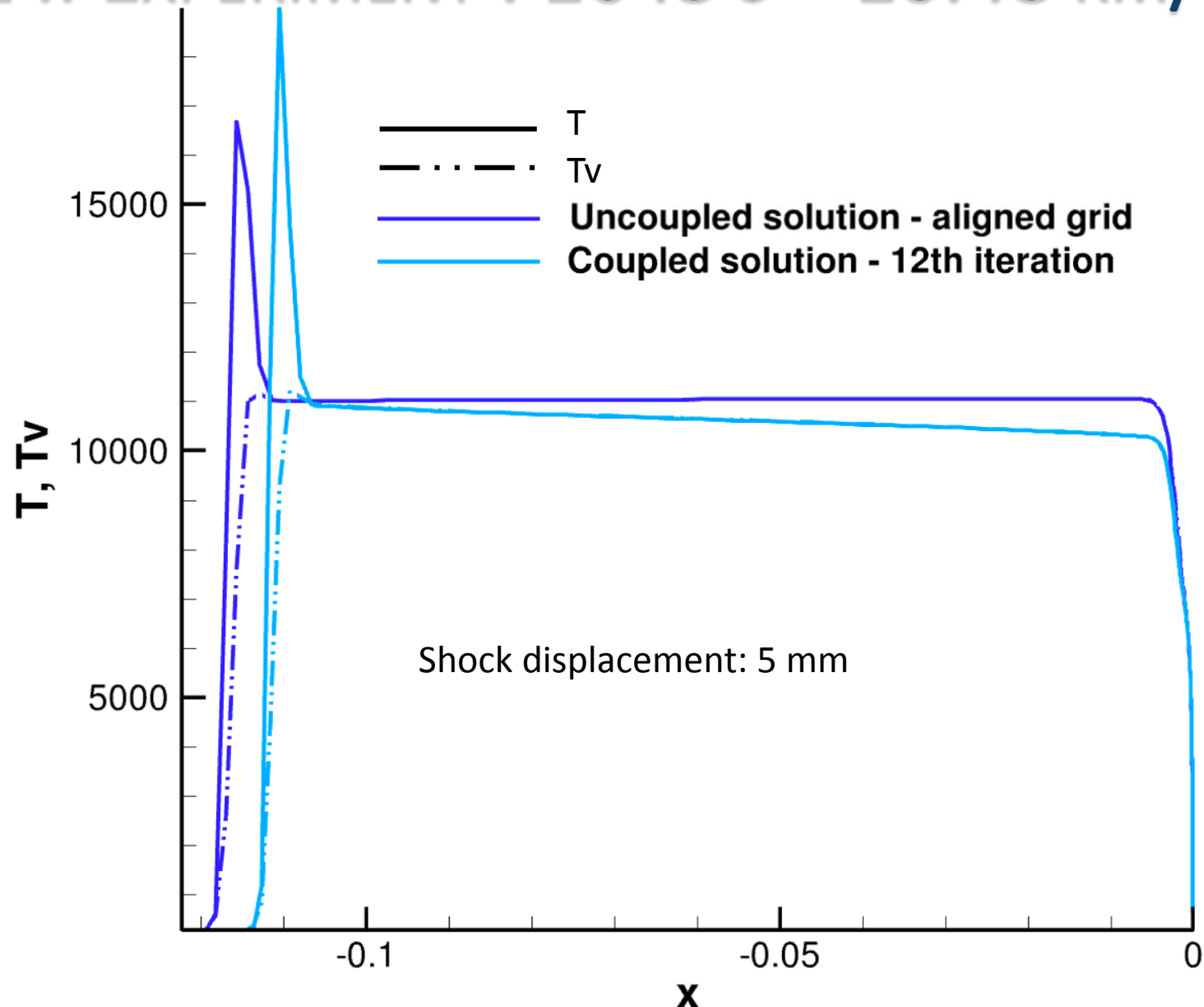
# FIRE II EXPERIMENT : 1645 s – 9.83 km/s



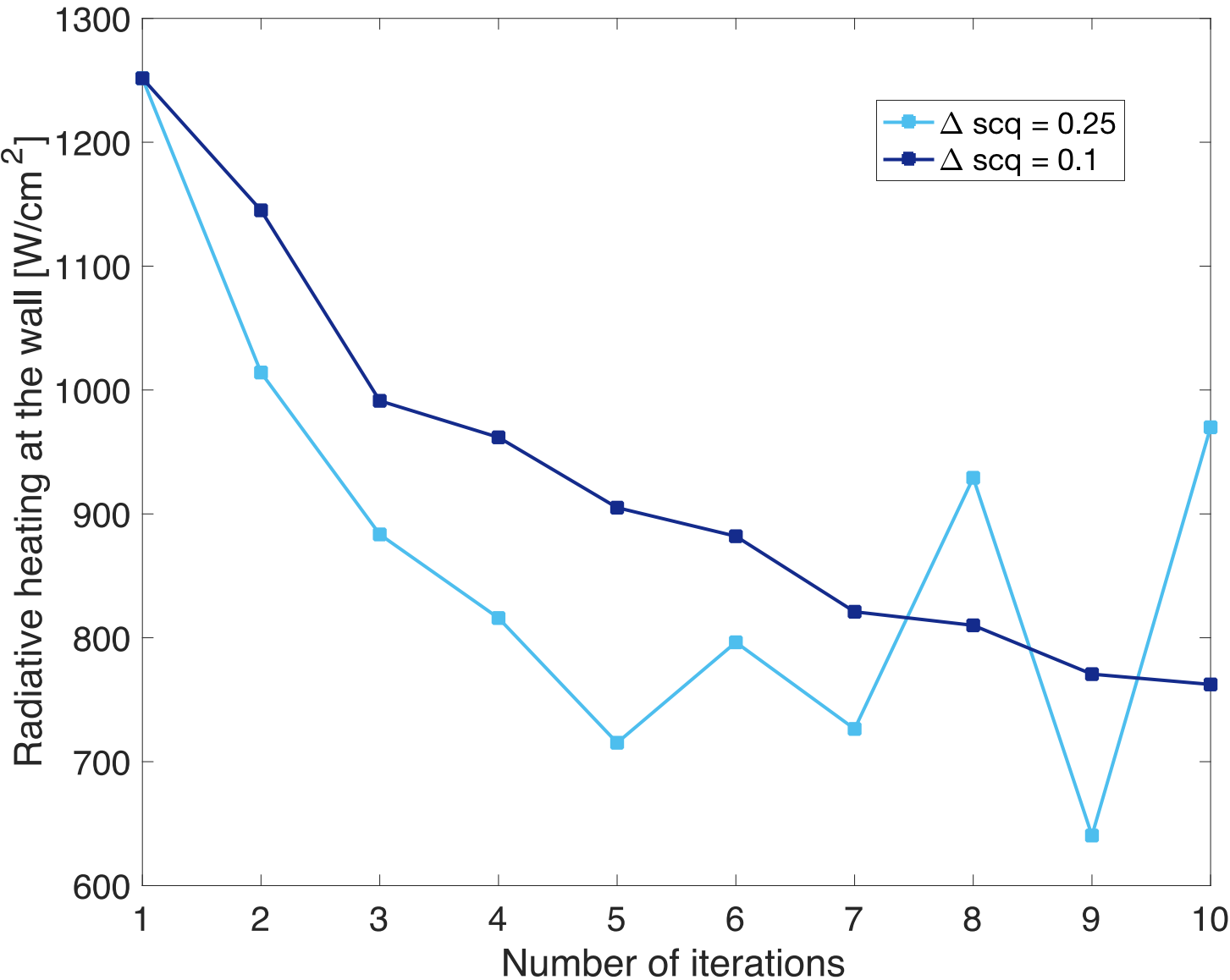
# FIRE II EXPERIMENT : 1645 s – 9.83 km/s



# FIRE II EXPERIMENT : 1643 s – 10.48 km/s



# FIRE II EXPERIMENT : 1643 s – 10.48 km/s



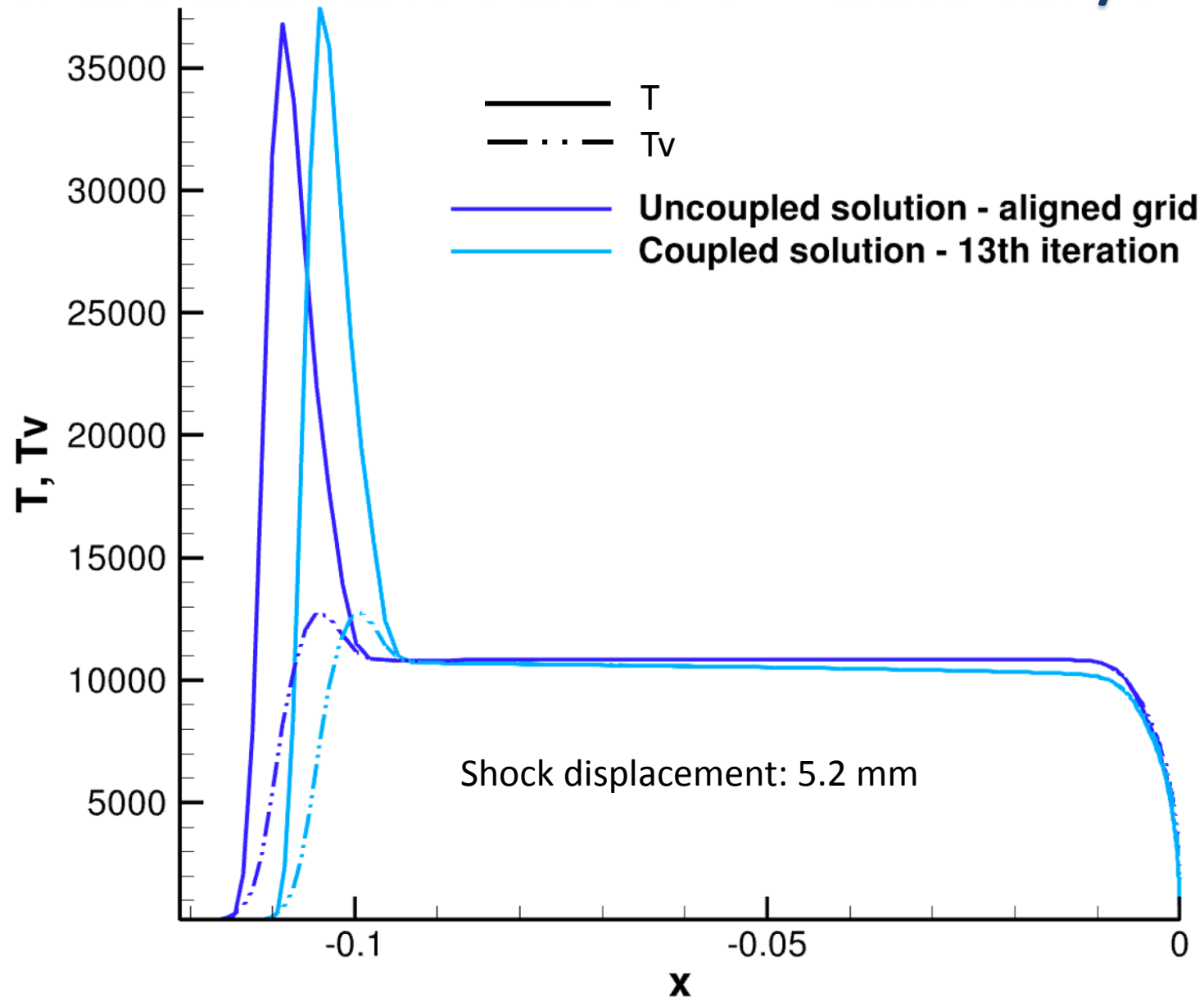
Applied scale factor:

- 1<sup>st</sup> it. →  $\text{scq}=0.25$
- 2<sup>nd</sup> it. →  $\text{scq}=0.5$
- 3<sup>rd</sup> it. →  $\text{scq}=0.75$
- 4<sup>th</sup>:10<sup>th</sup> its. →  $\text{scq}=1$

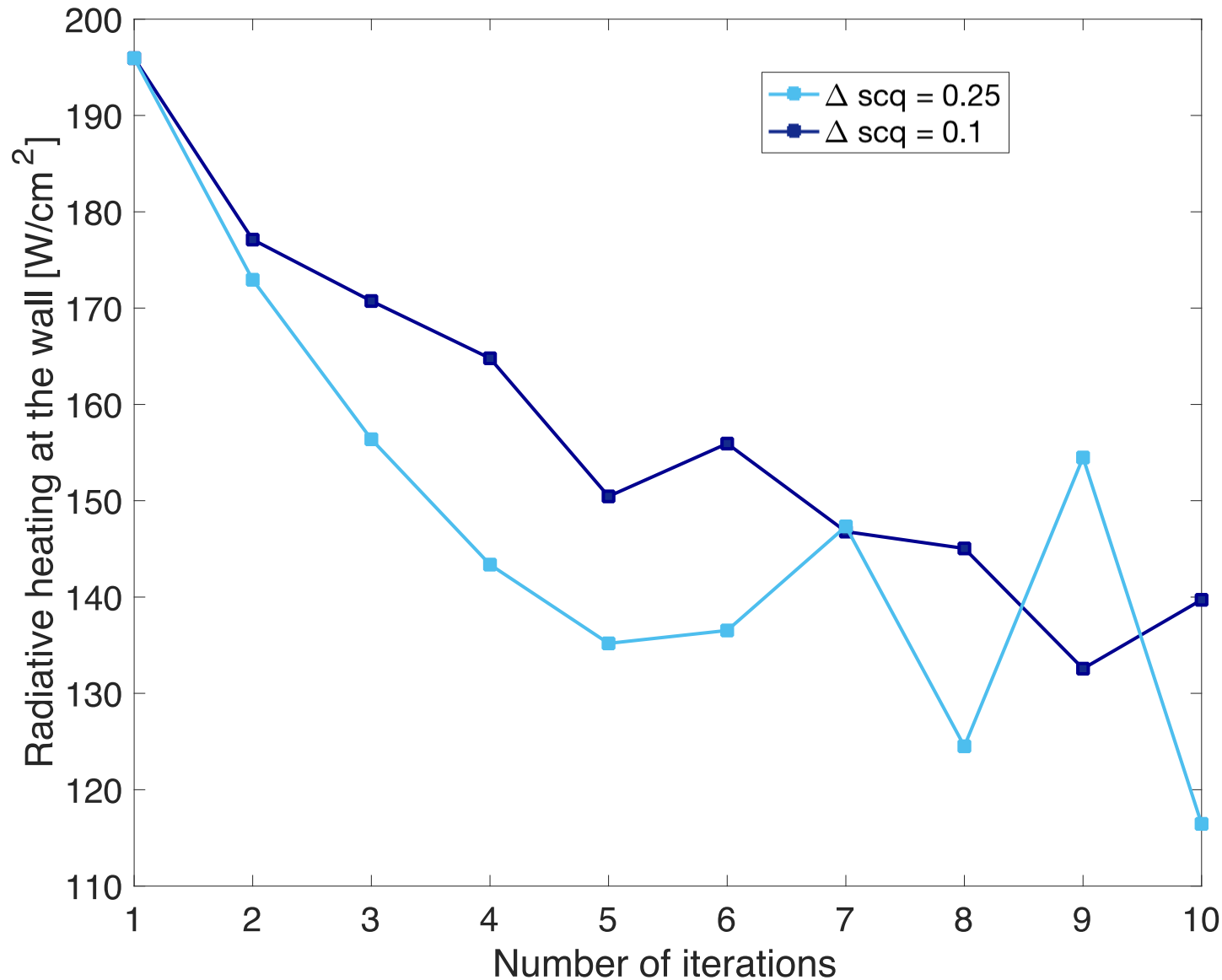
- 1<sup>st</sup> it. →  $\text{scq}=0.2$
- 2<sup>nd</sup> it. →  $\text{scq}=0.3$
- 3<sup>rd</sup> it. →  $\text{scq}=0.4$
- 4<sup>th</sup> it. →  $\text{scq}=0.5$
- 5<sup>th</sup> it. →  $\text{scq}=0.6$
- 6<sup>th</sup> it. →  $\text{scq}=0.7$
- 7<sup>th</sup> it. →  $\text{scq}=0.8$
- 8<sup>th</sup> it. →  $\text{scq}=0.9$
- 9<sup>th</sup>:10<sup>th</sup> its. →  $\text{scq}=1$



# FIRE II EXPERIMENT : 1636 s – 11.3 km/s



# FIRE II EXPERIMENT : 1636 s – 11.3 km/s



Applied scale factor:

- 1<sup>st</sup> it. →  $\text{scq}=0.25$
- 2<sup>nd</sup> it. →  $\text{scq}=0.5$
- 3<sup>rd</sup> it. →  $\text{scq}=0.75$
- 4<sup>th</sup>:10<sup>th</sup> its. →  $\text{scq}=1$

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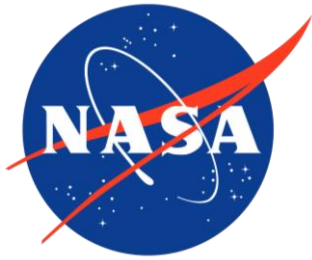
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# SUMMARY & FUTURE WORK

- Summer Achievements:
  - How the CFD analysis is influenced by different parameters, such as:
    - Grid alignment
    - Flow condition
  - First testing of the script for DPLR and NEQAIR coupling
    - Identifying bugs in the coupling code that were fixed during this work.
    - Now scripts functionally work, and will be soon ready for release
  - How the coupling solution convergence is influenced by:
    - Number of coupling iterations
    - Flow conditions
    - Scale factor
    - Grid alignment
- Future work:
  - Further efficiencies improvements can still be made in the coupling code
  - Ongoing investigation in the convergence of the radiation solution
  - Implementation of the interpolation method for the source term between the LOS
  - Mechanization of the coupler procedure



# THANK YOU FOR THE ATTENTION

## CHIARA AMATO